

*File
100*



J-23

IN-FLIGHT FAILURE ANALYSIS SUMMARY



Manager
Payload Integration

9/9/65
ats

Declassified and Released by the N R O

in Accordance with E. O. 12958

NOV 26 1997

SECRET

SUMMARY OF FAILURE

J-23 flight vehicle was operating normally for over one hundred orbits. However, during orbit No. 102 when the system was programmed to operate emergency mode (system on stereo and by-passing intermix circuits) the slave instrument completed 147 cycles as programmed while the master instrument only operated on the last 34 cycles of the program.

On orbit No. 104 (engineering pass) the master instrument failed to operate on stereo mode. Telemetry data had shown no drive voltage on the instrument which indicated the operate relays were not pulled in. At the same time the program on/off monitor telemetry data reflected the receipt by the operate command. After analyzing the T/M data it is concluded the probable cause of the failure lies between the holding relay for the master instrument operate signal in the command box, and the operate relays of the master instrument.

The master instrument failed to respond to operate signal in either mono 1 mode or stereo mode from orbit No. 104 through to orbit No. 132. Meanwhile no FMU TM changes were noted during any of the mono 1 mode operations.

On orbit No. 133 the master instrument operated on the last 2 cycles of 162 programmed cycles. On orbits No. 134 and 135 it responded to operate signals as programmed, but from then on until orbit No. 144, SRV-B recovery, it failed again to respond to any operate command.

On Orbit no. 183 the vehicle went through the deactivate exercise and the system responded as designed, both the master and slave instrument operated.

CONCLUSION

In reviewing the events, it is concluded that the most probable cause of failure is located in the command box on the circuit associated with the master instrument operate relay, K-13, which feeds the voltage to the FMU and the master instrument (ref. block diagram and schematic attached.)

When both components did not receive the necessary voltage, the suspicion is logically cast on K-13 circuitry. This further substantiated when the by-pass circuit, vehicle deactivate, successfully operated the system.

OPERATIONAL INVESTIGATIONS

As result of the above analysis, the following actions were taken to minimize failures of similiar nature on future systems.

A. Pad-readied System

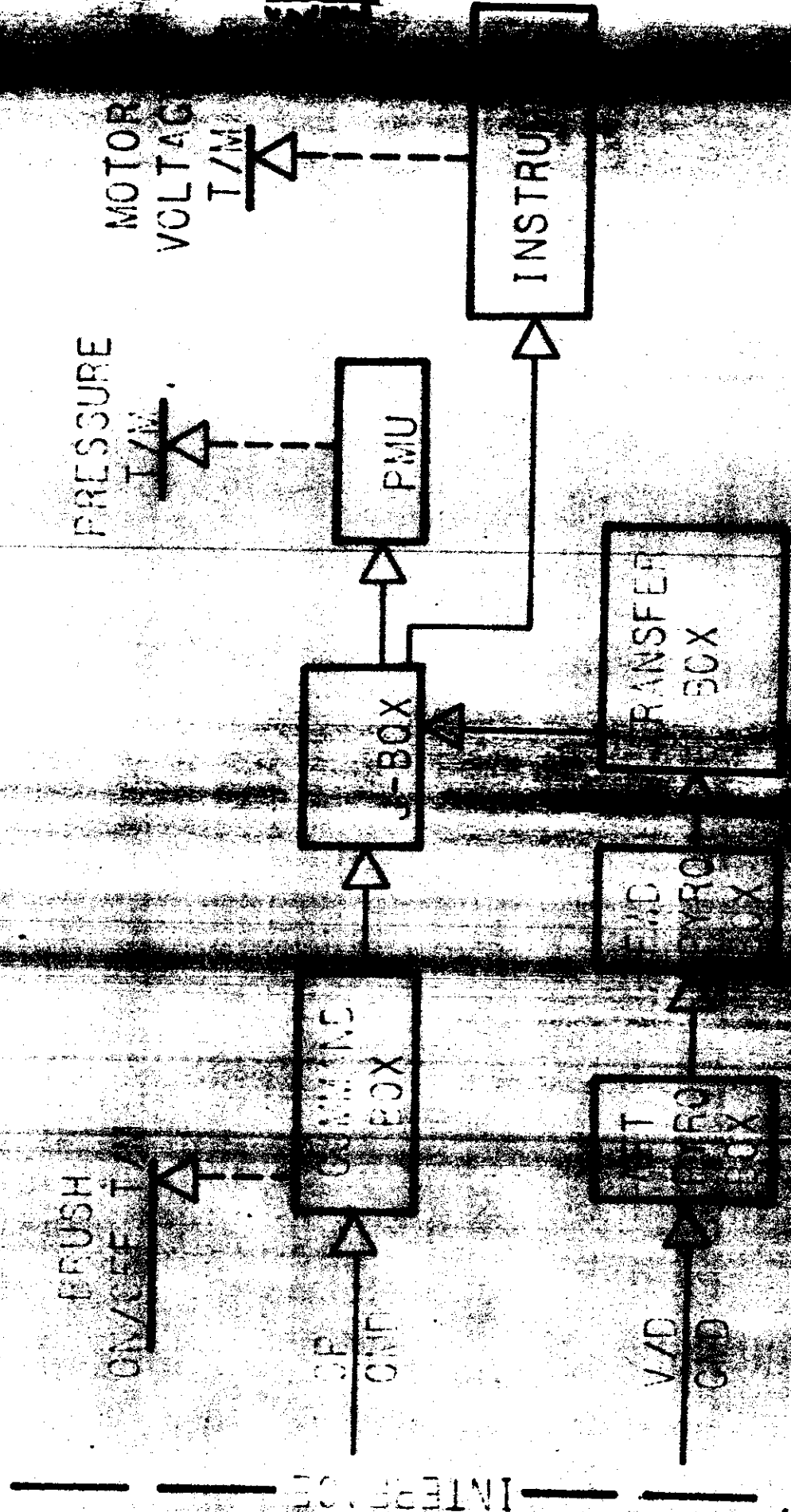
All Black-box assemblies from J-24 system were subjected to vibration and functional re-tests to assure integrity and to develop higher confidence levels of the holding relays and their associated circuitries. No anomalies were detected and all assemblies undergone tests successfully.

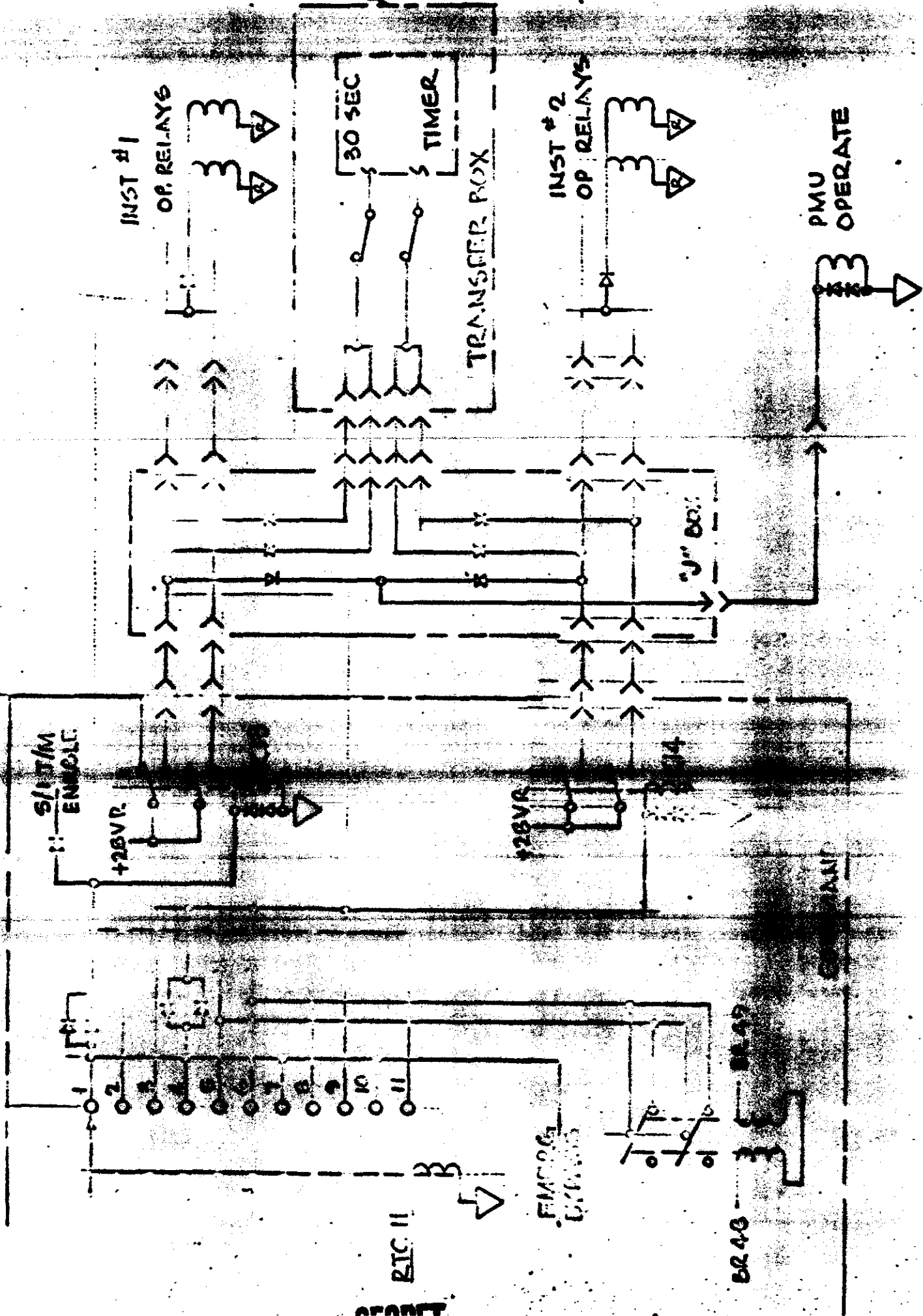
B. Future Systems

In the course of the failure analysis, potential problem areas in the system were uncovered and investigations are now underway to improve the condition. They are as follows:

1. CR-117 on the operate relay line in the instrument no longer serve its original function. Should it fail open or the connecting lead break the operation of the instrument will be permanently halted.
2. Operate relays in the command box feed operate voltages to the instruments for both standard operation mode and emergency operation mode. Any relay circuit failure as in the case of J23 will disable the instrument.

BLOCK DIAGRAM INSTRUMENT OPERATE CUM AND





12

RIC II



EMERGENCY

BRAG

BRAN

EMERGENCY

1-11-54 ILU

1. UNIT 1 (MOTOR)

A - TESTS AT FAILURE (MAY, 1952)

| | | | | | | | |
|-----|---|---|---|----|----|----|----|
| 112 | 6 | 8 | 9 | 10 | 11 | 12 | 15 |
| PCs | 7 | 5 | 1 | 6 | 7 | 8 | 4 |

B - FAILURE:

- POWERIX OFF

- OPERATING MODE

- FAILURE FILED 34 out of 147 STALLS GENERATED.

2. PASS 104 (MOTOR/AVE)

A - TESTS AT FAILURE ([REDACTED] 1952)

| | | | | | | | |
|-----|---|---|---|----|----|----|----|
| 112 | 6 | 8 | 9 | 10 | 11 | 12 | 15 |
| PCs | 7 | 5 | 3 | 6 | 7 | 11 | 4 |

B - FAILURE:

- INTERLOCK ON

- PROGRAM 3

- MASTER INSTRUMENT FAILED TO OPERATE;

Operate Relay monitor indicates to Operate signal received. Drive motor armature

U-23 FAILURE (Continued)

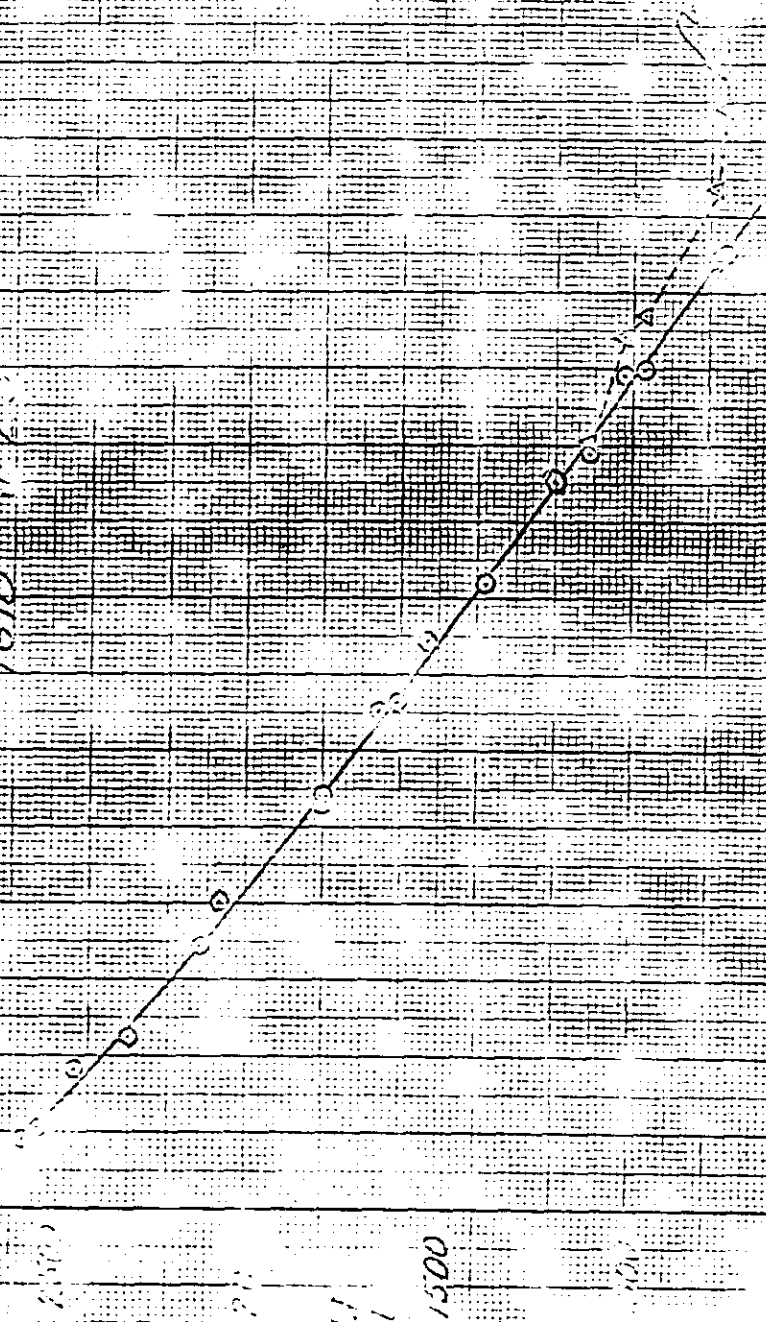
monitor read Zero volts. S/I Programmer
and S/I not working. Temp Sensors. Clock
and PMU were all OK.

3. DIAGNOSTIC OPERATIONS PERFORMED

- A - PASS 105 RTC 10 to Power on 1 (STEREO).
No operate on Pass 105.
- B - PASS 106 RTC 10 to Power on 2 (MONO #1).
No operate on Pass 106.

PMU GAS CONSUMPTION

1618 T-23



NO. BOTTLE
PRESSURE

1500

OPTIONAL OPERATE
WITH CALIBRATED
SCALE

100 200 300
ACCUMULATE TIME
MIN.

